Spring 2014

PHYS-2010

Lecture 23



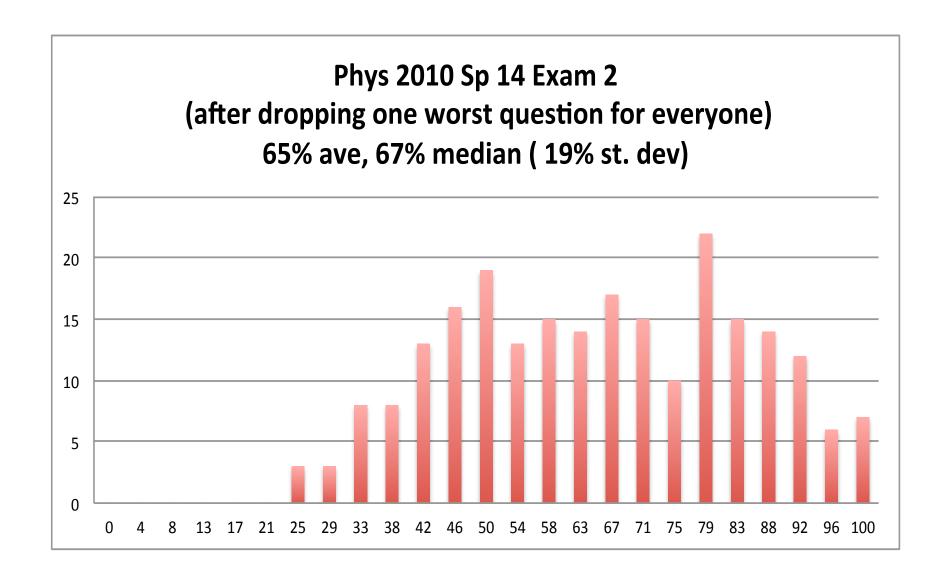
Astronauts aboard the International Space Station float around, experiencing weightlessness.

Why is this?

- A) The force of gravity from the earth is zero on the Space Station
- B) The force of gravity is much, much weaker on the Space Station
- C) The Space Station has the "inertial dampers" turned on.
- D) The Space Station is in circular orbit around the earth.
- E) The Space Station generates an anti-gravity field.

Announcements

- Finish Giancoli Chapter 5, start Chapter 6.
- No Written homework this week!
- CAPA # 8 is due next Tuesday.
- Exam solutions are posted on D2L.
- Exam results will be posted by the weekend's end.





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Newton's 2nd law and law of gravity:

$$mg = F_g = G \frac{mM}{R_E^2} \rightarrow g = a_g = G \frac{M}{R_E^2}$$

Gravity at the surface of the Earth:

$$a = \frac{GM_E}{R_E^2} = \left(6.67 \times 10^{-11} Nm^2 / kg^2\right) \frac{\left(5.98 \times 10^{24} kg\right)}{\left(6.37 \times 10^6 m\right)^2} = 9.81 m/s^2$$

Gravity at the Space Station Orbit above the Earth h ~ 300 km = 3 x 10⁵ meters:

$$a = \frac{GM_E}{\left(R_E + h\right)^2} = \left(6.67 \times 10^{-11} Nm^2 / kg^2\right) \frac{\left(5.98 \times 10^{24} kg\right)}{\left(6.37 \times 10^6 m + 3 \times 10^5 m\right)^2}$$

$$= 8.43 m/s^2$$

Gravitational acceleration is a little weaker, but not so much.

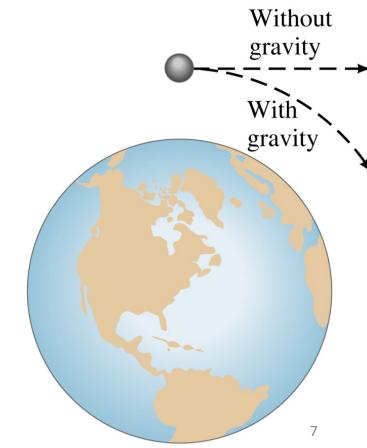
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Satellites and "Weightlessness"

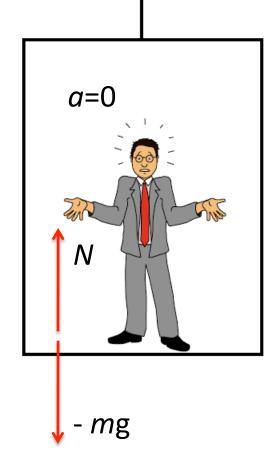
Satellites are routinely put into orbit around the Earth. The tangential speed must be high enough so that the satellite does not return to Earth, and not so high that it escapes Earth's gravity altogether.

The satellite is kept in orbit by its speed – it is continually falling, but the Earth curves from underneath it.

Because of its continual falling, it is considered to be "weightless".



"Weightlessness" and "Free Fall"



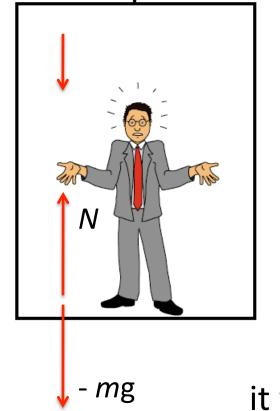
$$F_{net} = ma = N - mg = 0$$

$$N = mg$$

If the person were on a scale, it would read their regular weight.

"Weightlessness" and "Free Fall"

-a (controlled fall)



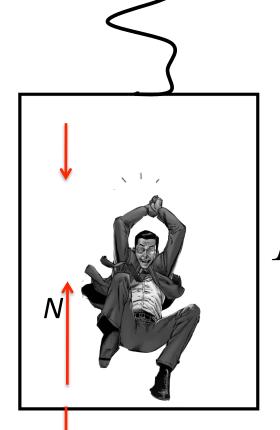
$$F_{net} = ma = N - mg \neq 0$$

$$N = m(g - a) < mg$$

If the person were on a scale, it would read less than their regular weight.

"Weightlessness" and "Free Fall"

-g (free fall)



Therefore, when gravity alone is operating unopposed by another force (e.g., normal force), the object is said to be **weightless**.

The object is also said to be in **free fall**.

$$N = m(g - a) = m(g - g) = 0$$

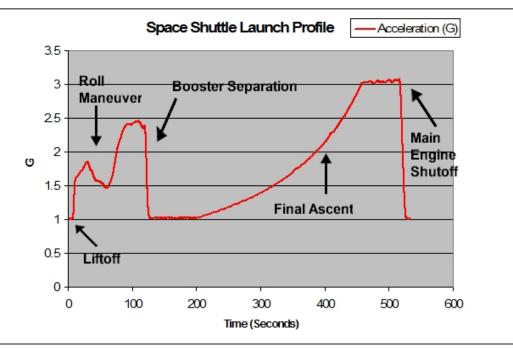
Weightless!

If the person were on a scale, it would read zero.

Clicker Question

Room Frequency BA





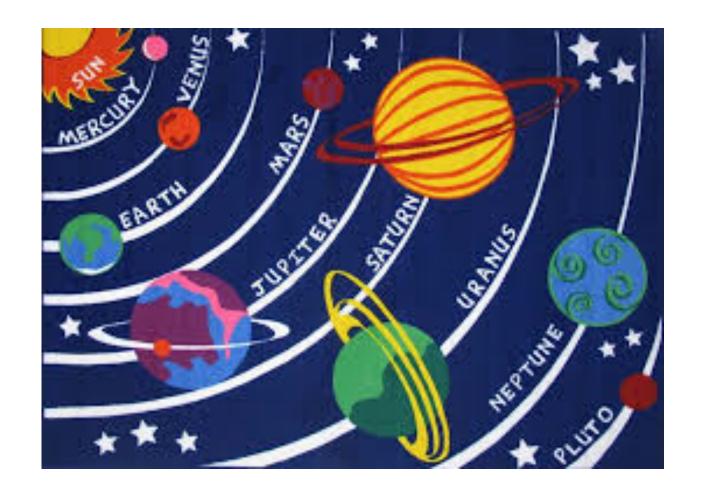
What happens to a person's apparent "weight" as measured on a scale on the shuttle at t=100 seconds, compared to that on Earth?

A) Stays the same

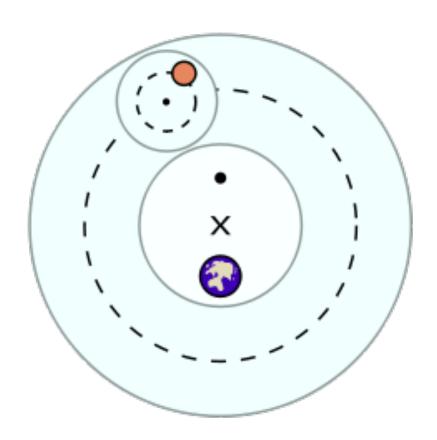
B) Increases

C) Decreases

ORBITS AND KEPLER'S LAWS



Earth at the Center of Everything





- Epicyles
- Equant
- Deferent



Claudius Ptolemaeus (Ptolemy)

90 - 168 AD

Heliocentrism (Sun at the Center)



Nicolaus Copernicus 1473-1543

De revolutionibus orbium coelestium (On the Revolutions of the Celestial Spheres)

Heliocentrism



Tycho Brahe 1546-1601

astronomical observations



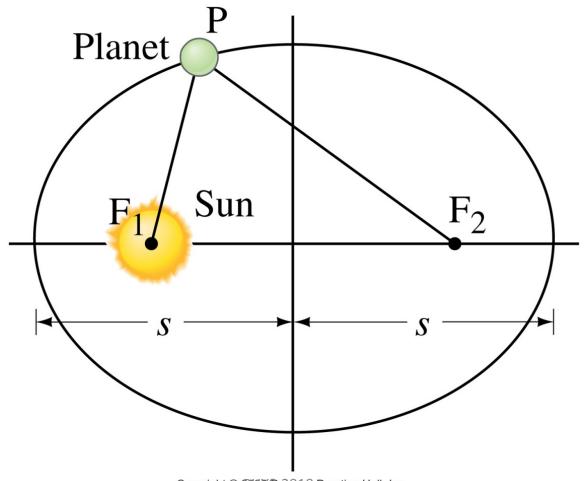
Johannes Kepler 1571-1630

Three Laws

Epitome astronomia Copernicanae (Epitome of Copernican Astronomy)

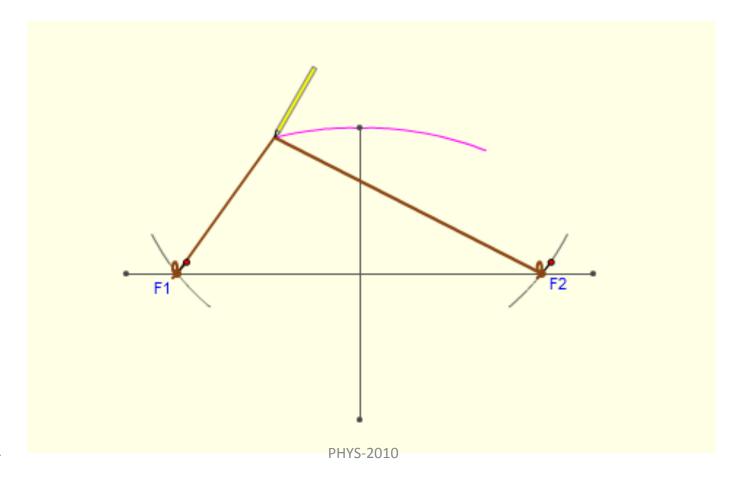
Kepler's Laws of Planetary Motion

Law #1: The orbit of each planet is an ellipse, with the Sun at one focus.



Elliptical Geometry

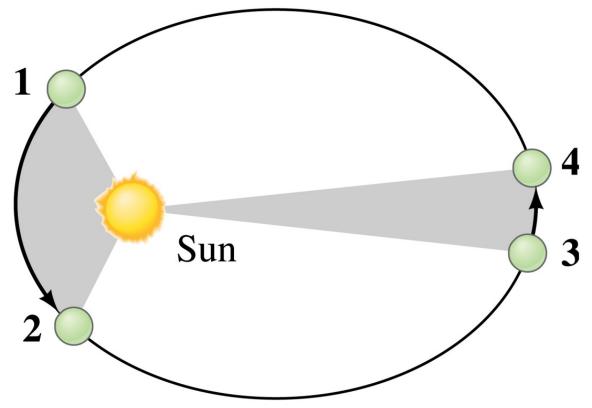
The sum of the distances from any point P on the ellipse to those two foci is constant and equal to the major diameter $(PF_1 + PF_2 = 2a)$.



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Kepler's Laws of Planetary Motion

Law#2: An imaginary line drawn from each planet to the Sun sweeps out equal areas in equal times.



Notice how a planet with an elliptical orbit moves closer to and further away from the Sun. The point of closest approach to the Sun is called **perihelion**. The furthest point is called **aphelion**.



At which position is the asteroid moving at the largest magnitude velocity?

- A) Far left (aphelion)
- B) Far right (perihelion)
 - C) Top
 - D) Bottom
- E) Same velocity magnitude everywhere

Kepler's Laws of Planetary Motion

Law #3: The ratio of the square of a planet's orbital period is proportional to the cube of its mean distance from the Sun.

$$T^2 \propto R^3$$

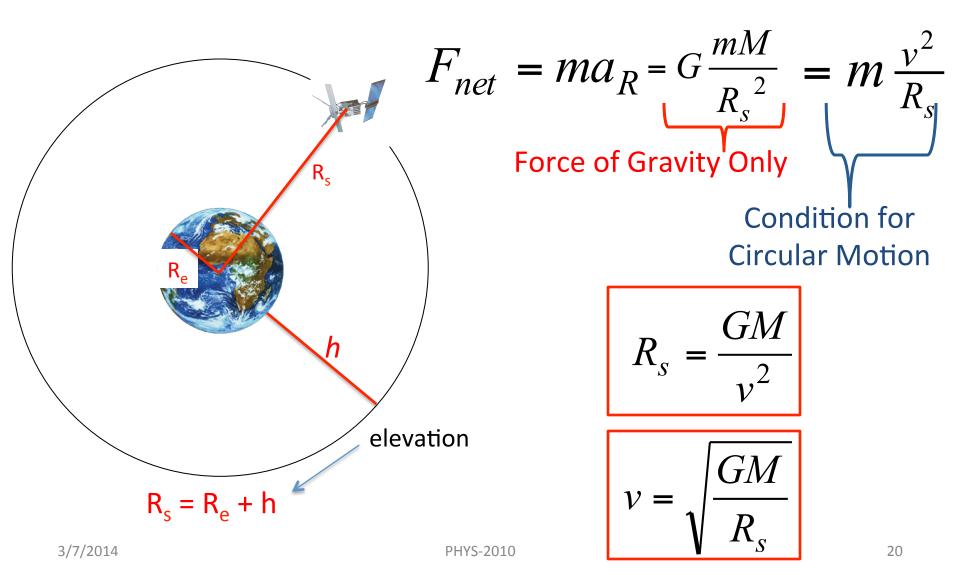
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$$\frac{R^3}{T^2} = const$$

TABLE 5-2	Planetary Data Applie			
Planet	Mean Distance from Sun, <i>s</i> (10 ⁶ km)	Period, <i>T</i> (Earth years)	s^3/T^2 $(10^{24} \mathrm{km}^3/\mathrm{y}^2)$	
Mercury	57.9	0.241	3.34	
Venus	108.2			
Earth	149.6	2		
Mars	227.9	$\mathbf{R}^{\mathfrak{I}}$		
Jupiter	778.3		2 2 5 1 6	$)^{24} km^3 / yr^2$
Saturn	1427		3.33×10	$)^{-1}KM^{\circ}/Vr^{-1}$
Uranus	2870	$\mathbf{I}_{\mathbf{m}}^{2}$, , , , ,
Neptune	4497	<i> </i>		
Pluto	5900	210	J.J.1	
3/7/2014	Convright © 2005 P	earson Prentice Hall, Inc. 1945-2	010	19

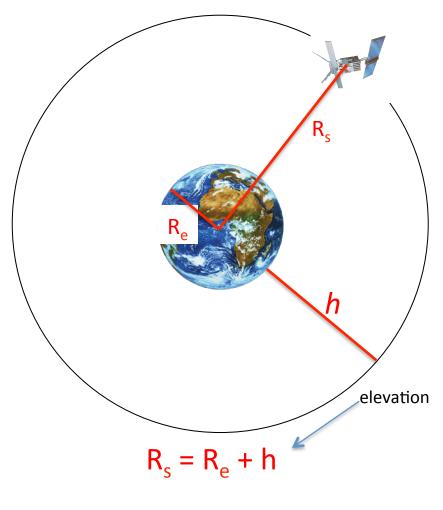
What is the relationship between velocity and orbit Radius?

Consider F = ma in the radial direction: +inward



Clicker Question

Room Frequency BA



$$v = \sqrt{\frac{GM}{R_s}} \qquad R_s = \frac{GM}{v^2}$$

Satellite 1 is in circular orbit about the Earth and is moving with tangential speed v_1 .

Satellite 2 is also in orbit but with speed $v_2 > v_1$.

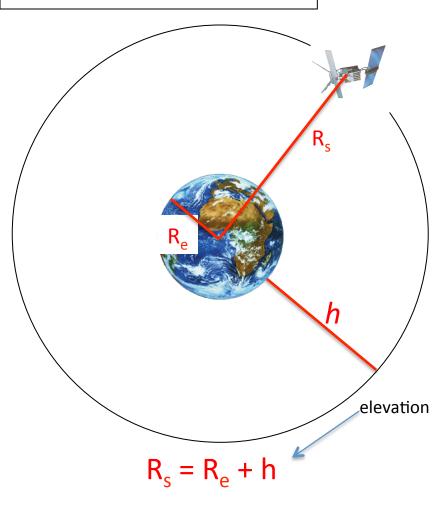
Satellite 2's orbit radius is:

- A) Bigger than Satellite 1's
- B) Smaller than Satellite 1's
- C) Unclear from the information given.

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Clicker Question

Room Frequency BA



$$v = \sqrt{\frac{GM}{R_s}} \qquad R_s = \frac{GM}{v^2}$$

Satellite 1 is in circular orbit about the Earth with radius R_1 .

Satellite 2 is also in circular orbit with a radius R_2 and a speed twice that of Satellite 1.

What is true about the radii?

- A) $R1 = 2 \times R2$
- B) $R1 = \frac{1}{2} \times R2$
- C) $R1 = 4 \times R2$
- D) $R1 = \frac{1}{4} \times R2$
- E) Unclear from the information given.

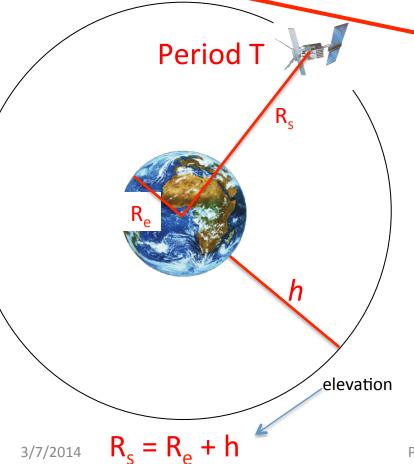
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Relationship between orbit radius and period

$$v = \sqrt{\frac{GM}{R_s}} \qquad R_s = \frac{GM}{v^2}$$

Eliminate velocity in place of period (T)

$$v = \frac{2\pi R_s}{T}$$



$$v^2 = \frac{GM}{R_s} = \frac{4\pi^2 R_s^2}{T^2}$$

$$\frac{R_s^3}{T^2} = \frac{GM}{4\pi^2}$$

$$\frac{GM}{R_s} = \frac{4\pi^2 R_s^2}{T^2} \qquad \frac{R_s^3}{T^2} = \frac{GM}{4\pi^2}$$

to Kepler's Third Law

Period, T

TABLE 5-2 Planetary Data Applied					
Planet	Mean Distance from Sun, <i>s</i> (10 ⁶ km)				
Mercury	57.9				
Venus	108.2				
Earth	149.6				
Mars	227.9				
Jupiter	778.3				
Saturn	1427				
Uranus	2870				
Neptune	4497				
Pluto	5900				
	Copyright © 2005 Pea				

Recall Kepler's Empirical Value

$$\frac{R^3}{T^2} = 3.35 \times 10^{24} \, km^3 \, / \, yr^2$$

$$\frac{R_s^3}{T^2} = \frac{GM}{4\pi^2} = \frac{\left(6.67 \times 10^{-11} Nm^2 / kg^2\right) \left(1.98 \times 10^{30} kg\right)}{4\pi^2} = 3.3 \times 10^{18} m^3 / s^2$$

$$= 3.3 \times 10^{24} km^3 / yr^2$$
PHYS-2010
$$= 3.3 \times 10^{24} km^3 / yr^2$$

Knowledge of a planet's period (easy to measure) allows us to determine it orbital radius.

Where is Mars?

$$\frac{R_s^3}{T^2} = \frac{GM}{4\pi^2} = \text{constant}$$

$$\frac{R_{sun-mars}^3}{T_{mars}^2} = \frac{R_{sun-earth}^3}{T_{earth}^2} = \text{constant}$$

$$\frac{R_{sun-mars}^{3}}{R_{sun-earth}^{3}} = \frac{T_{mars}^{2}}{T_{earth}^{2}}$$



Mars is 1.52 times as far away as the earth is from the sun.

$$\frac{R_{sun-mars}}{R_{sun-earth}^{3/7/2014}} = \left(\frac{T_{mars}}{T_{earth}}\right)^{2/3} = \left(\frac{687 days}{365 days}\right)^{2/3} = 1.52$$

Exoplanets

Upsilon Andromedae is a binary star located about 44 light-years away from the Earth. The primary star is a yellow-white dwarf star that is younger than the Sun. There is a second star that is a red dwarf in a wide orbit.

As of 2010, four confirmed extrasolar planets have been discovered.

The Upsilon Andromedae A system^[23]

Companion (in order from star)	Mass	Semimajor axis (AU)	Orbital period (days)	Eccentricity
b	1.4 ^[24] <i>M</i> J	0.0595 ± 0.0034	4.617136 ± 0.000047	0.013 ± 0.016
С	13.98 ^[24] <i>M</i> J	0.832 ± 0.048	241.33 ± 0.20	0.224 ± 0.021
d	10.25 ^[24] <i>M</i> J	2.53 ± 0.15	1278.1 ± 2.9	0.267 ± 0.021
е	≥1.059 ± 0.028 ^[21] M _J	5.2456 ± 0.00067	3848.86 ± 0.74	0.00536 ± 0.00044

$$\frac{R_s^3}{T^2} =$$

(b)
$$\rightarrow$$
 (0.0595)³/(4.61)²=9.9x10⁻⁶ AU³/days²

(c)
$$\rightarrow$$
 (0.832)³/(241.3)²=9.9x10⁻⁶ AU³/days²

(d)
$$\rightarrow$$
 (2.53)³/(1278.1)²=9.9x10⁻⁶ AU³/days²

(e)
$$\rightarrow$$
 (5.24)³/(3848₅8)²=9.7x10⁻⁶ AU³/days²